弹性支撑浅拱的非线性动力行为分析*

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摘要本文研究了两端转角均为转动弹簧支撑的铰支浅拱在外激励作用下的非线性动力学行为.基于弹性 支撑浅拱的基本动力控制方程,采用多尺度法对内共振进行了摄动分析,并得到了极坐标形式的平均方程. 弹性约束的刚度通过特征方程影响结构的自振频率和模态,且与平均方程的相关系数一一对应,文中还以 最低两阶模态之间1:1内共振为对象进行了数值分析.结果显示系统存在模态交叉与转向两种内共振形 式,另一方面结构参数处于某一范围之内时外激励激发的模态作用可导致出现准周期运动和混沌运动.

关键词 浅拱, 转动弹性支撑, 内共振, 分岔, 模态转向

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引 言

拱结构^[1]线形优美、受力性能良好,有着广泛的 工程背景.国内外学者对各种外荷载作用下的静、动 力学进行了深入的研究,涉及到几何缺陷[2]、承载能 力^[3-5]、跳跃屈曲^[6-7]及分岔混沌^[8-10]等各个方面. 这些研究中,拱的边界一般假定为理想的固支或铰 支,实际上在工程实践中拱的边界在某些情况下不 能简单地视为铰支或固支,如柔性系杆拱中存在系 杆使得其力学模型可抽象为水平弹性支撑拱;另外 一些机械拱臂或曲臂是存在平动和转动的弹性约 束.对于弹性和粘弹性边界结构的研究主要集中于 梁[11-12],较少涉及到拱的动力行为.由于拱结构与 相邻结构共同承受载荷,动力荷载下基础变形引起 附加惯性力,将对结构的动力响应产生影响,因此采 用弹性支撑边界来研究结构的动力行为更加合理. 本文以两端转动弹性约束浅拱为研究对象,采用多 尺度法[13]分析最低两阶模态之间的1:1内共振现 象及相应的分岔、混沌等非线性动力学行为.

1 分析模型

1.1 基本方程

图 1 所示直角坐标系 $\delta - \hat{x}$ 下跨径为的两端 转动弹性支撑浅拱,弹簧刚度分别为 \hat{k}_1 和 \hat{k}_2 , \hat{y}_0 (\hat{x})为初始时刻的拱轴线, $\hat{y}(\hat{x}, i)$ 为i时刻在外荷载 $f(\hat{x}, i)$ 作用下的位置.引入基本假定(1)平截面假定;(2)不考虑剪切变形和转动惯量;(3)零初始轴向力,动力学控制方程可写为^[13]:



图1 转动弹性支撑浅拱结构示意图

Fig. 1 The schematic torsional spring support shallow arch and the excitation

$$\rho A \frac{\partial^2 \hat{y}}{\partial t^2} + EI \frac{\partial^4 \hat{y}}{\partial \hat{x}^4} - \frac{EA}{l} \frac{d^2 \hat{y}_0}{d \hat{x}^2} \int_0^l \frac{\partial \hat{y}}{\partial \hat{x}} \frac{d \hat{y}}{d \hat{x}} d\hat{x} = \frac{EA}{l} \frac{\partial^2 \hat{y}}{\partial \hat{x}^2} \int_0^l \frac{\partial \hat{y}}{\partial \hat{x}} \frac{d \hat{y}_0}{d \hat{x}} d\hat{x} + \frac{EA}{2l} \frac{d^2 \hat{y}_0}{d \hat{x}^2} \int_0^l (\frac{\partial \hat{y}}{\partial \hat{x}})^2 d\hat{x} + \frac{EA}{2l} \frac{\partial^2 \hat{y}_0}{\partial \hat{x}^2} \int_0^l (\frac{\partial \hat{y}}{\partial \hat{x}})^2 d\hat{x} - \hat{c} \frac{\partial^2 \hat{y}}{\partial \hat{t}} - f(\hat{x}, \hat{t})$$
(1)

其中 A 为截面积, I 为转动惯量, ρ 为密度, E 为弹 性模量, \hat{c} 为阻尼系数, 边界条件为:

$$\hat{x} = 0; \quad \hat{y} = 0, EI \frac{\partial^2 \hat{y}}{\partial \hat{x}^2} - \hat{k}_1 \frac{\partial \hat{y}}{\partial \hat{x}} = 0 \quad (2)$$

$$\hat{x} = l: \quad \hat{y} = 0, EI \frac{\partial^2 \hat{y}}{\partial \hat{x}^2} + \hat{k}_2 \frac{\partial \hat{y}}{\partial \hat{x}} = 0$$
(3)

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引入如下无量纲参变量

$$x = \frac{\hat{x}}{l}$$
; $\psi = \frac{\hat{y}_0}{r}$; $u = \frac{\hat{y}}{r}$; $t = t \sqrt{\frac{EI}{\rho A l^4}}$ (4)

其中r是截面的转动惯量,方程可无量纲化如下

$$\frac{\partial^2 u}{\partial t^2} + \frac{\partial^4 u}{\partial x^4} - \frac{d^2 \psi}{dx^2} \int_0^1 \frac{\partial u}{\partial x} \frac{d\psi}{dx} dx = \frac{\partial^2 u}{\partial t^2} \int_0^1 \frac{\partial u}{\partial x} \frac{d\psi}{dx} dx + \frac{1}{2} \frac{d^2 \psi}{dx^2} \int_0^1 (\frac{\partial u}{\partial x})^2 dx + \frac{1}{2} \frac{\partial^2 u}{\partial t^2} \int_0^1 (\frac{\partial u}{\partial x})^2 dx - \varepsilon^2 c \frac{\partial u}{\partial t} - \varepsilon^3 F \cos \Omega t$$
(5)

式中 ε 是无量纲小参数, $\varepsilon^2 c = l^2 / \sqrt{\rho A E I \hat{c}}$, $\varepsilon^3 F \cos \Omega t$ = $(l^4 / E h) f(\hat{x}, \hat{t})$ 为谐波激励, $\varepsilon^3 F$ 是其幅值, 方程 对应的边界为

$$u = 0, \ \frac{\partial^2 u}{\partial x^2} - k_1 \ \frac{\partial u}{\partial x} = 0 \ \text{at} \quad x = 0$$
 (6)

$$u = 0, \ \frac{\partial^2 u}{\partial x^2} + k_1 \ \frac{\partial u}{\partial x} = 0 \ \text{at} \quad x = 1$$
 (7)

其中 $k_1 = \hat{k}_1 l/EI$ 和 $k_2 = \hat{k}_2 l/EI$ 是无量纲转动弹簧 刚度,并在后续摄动分析中假定为O(1).

1.2 自振特性

将式(5)中阻尼、外荷载和非线性项去掉,可 得用于分析结构自振频率和模态的方程

$$\frac{\partial^2 u}{\partial t^2} + Lu = \frac{\partial^2 u}{\partial t^2} + \frac{\partial^4 u}{\partial x^4} - \frac{d^2 \psi}{dx^2} \int_0^1 \frac{\partial u}{\partial x} \frac{d\psi}{dx} dx = 0$$
(8)

边界条件仍为(6)和(7),上式中是为了计算方便 引入的正线性自伴微分算子. *n* 阶频率 ω_n 对应正 交模态 $\phi_n(x)$,将其标准化有 $\int_0^1 \phi_m \phi_n dx = <\phi_m, \phi_n$ > = $\delta_{mn}(\delta$ 函数), $<\phi_m, L\phi_n > = \omega_n^2 \delta_{mn}$. 无量纲浅拱 方程可假定为 $\psi(x) = b \sin \psi x, b$ 为矢高,正交模态 可以表示为

 $\phi_n(x) = c_1 \cos \omega_n^{1/2} x + c_2 \sin \omega_n^{1/2} x + c_3 \cosh \omega_n^{1/2} x + c_4 \sinh \omega_n^{1/2} x + c_5 \sin \pi x$ (9)

其中 c_i 满足(6-8)式的特征方程的系数, ω_n 是特征解,详细表达式见附录(A1 - A5)式.

为考察结构参数变化时自振频率的分布规律, 图 2 给出了转动约束弹簧刚度 k_1 和 k_2 取两组不同 值时各阶频率随矢高 b 的变化图.从中可看到,相 邻阶次模态之间可能发生模态交叉(cross)或转向 (veer)的1:1内共振,且(a) k_1 = 10, k_2 = 10 时两种 形式在各递增的相邻模态间交替出现,(b) k_1 = 10, k₂ = 20 时各相邻模态间只有模态转向,这与两端铰 支或固支^[14]的模态形式存在明显差异.





图 3 最低两阶频率随(a)矢高和(b)弹簧刚度的变化示意图 Fig. 3 Variations of the first two frequencieswith (a) arch riseand (b) elastic stiffness

由图 2 可知 k₁ = 10 而 k₂ 取不同值时,1:1 内 共振的出现形式完全不同目对应参数存在差异,为 进一步研究转动弹簧刚度的影响,图3(a)以最低 两阶模态间的1:1内共振为研究对象给出了 k₁和 k,取几组不同值时随变化的分布规律,图中5组数 据显示当 $k_1 \neq k_2$ 时两阶模态之间出现转向, $k_1 = k_2$ =10 时则出现交叉.结合图2中的结果,这说明当 两端转动弹簧刚度相同结构完全对称时可能出现 模态交叉,而两端弹簧刚度不同结构不对称时则出 现模态转向. 图 3(b)则给出了 k1 从 0 到 20 以公差 5 递增的5 组数据中 ω 随 k_2 变化的示意图,图中b 取相应 k_1 和 k_2 =10时最低两阶模态发生1:1内共 振的值,即b分别等于6.1,6.5,6.7,6.9和7.0,从 图中可发现在 k2 增至一定值时频率值趋于稳定, 两阶模态的频率值除 $k_1 = 10, b = 6.7$ 这组数据在 k2 = 10 附近接近外均有不同大小的差值,这说明了 内共振以模态转向形式为主.

图 4 则给出了 k₁ = 10,k₂ = 20 时最低两阶频率 在转向附近的正交模态分布示意图,可发现模态既 非正对称也非反对称而是由二者组合而成,当 b 小 于模态转向值 7.0 时 1 阶模态以对称成分为主,2 阶 模态以反对称成分为主,当 b > 7.0 时则刚好相反.



图 4 最低两阶频率在转向位置附近的模态分布:

- (a) = 6.8, (b) = 6.9, (c) = 7.0, (d) = 7.1, (e) = 7.2
- Fig. 4 Variations of the first two frequencies along with the associated mode shapes with arch rise: (a) = 6.8, (b) = 6.9,
 - (c) =7.0, (d) =7.1 and (e) =7.2

2 内共振的摄动分析

式中上标点表示对 t 的微分, G_2 和 G_3 分别为二次、 三次非可换非线性微分算子, 其表达式为 $G_2(u,\nu)$ = $u'' < \nu', \psi' > + (1/2)\psi'' < u', \nu' > , G_3(u,\nu,w) =$ (1/2) < $\nu', w' >$, 上标撇表示对 x 的微分. 采用多 尺度法对其进行分解, 将 u 和 ν 的解直接一致展开 为

$$\begin{cases} u(x,t) = \varepsilon u_1(x,T_0,T_1,T_2) + \\ \varepsilon^2 u_2(x,T_0,T_1,T_2) + \varepsilon^3 u_3(x,T_0,T_1,T_2) + \cdots \\ \nu(x,t) = \varepsilon \nu_1(x,T_0,T_1,T_2) + \\ \varepsilon^2 \nu_2(x,T_0,T_1,T_2) + \varepsilon^3 \nu_3(x,T_0,T_1,T_2) + \cdots \end{cases}$$
(11)

式中时间 $T_0 = t$, $T_1 = \varepsilon t$, $T_2 = \varepsilon^2 t$, 且有 $\partial/\partial t = D_0 + \varepsilon^1 D_1 + \varepsilon^2 D_2 + \cdots$, $D_n = \partial/\partial T_n$. 将其代入(10) 式展 开并利用时间参数的独立性可得近似方程如下

$$\varepsilon: \begin{cases} D_0 u_1 - \nu_1 = 0\\ D_0 \nu_1 + L u_1 = 0 \end{cases}$$
(12)

$$\varepsilon^{2}: \begin{cases} D_{0}u_{2} - \nu_{2} = -D_{1}u_{1} \\ D_{1}u_{1} + U_{2} = -D_{1}u_{1} \\ (13) \end{cases}$$

$$\varepsilon^{3}: \begin{cases} D_{0}\nu_{2} + Lu_{2} = -D_{1}\nu_{1} + G_{2}(u_{1}, u_{1}) \\ D_{0}\nu_{3} - \nu_{3} = -D_{2}u_{1} - D_{1}u_{2} \\ D_{0}\nu_{3} + Lu_{3} = -D_{2}\nu_{1} - D_{1}\nu_{2} + \\ G_{2}(u_{1}, u_{2}) + G_{2}(u_{2}, u_{1}) + \\ G_{3}(u_{1}, u_{1}, u_{1}) - c\nu_{1} - F\cos\Omega t \end{cases}$$
(14)

上述方程组对应的边界条件均为(6)和(7),对于 m和n阶模态之间的相互作用,内共振只与这两阶 模态有关,因此可以将一阶近似方程(12)的解表 示为

$$\begin{cases} u_{1} = A_{m}(T_{1}, T_{2}) e^{i\omega_{m}T_{0}} \phi_{m}(x) + \\ A_{n}(T_{1}, T_{2}) e^{i\omega_{n}T_{0}} \phi_{n}(x) + cc \\ \nu_{1} = i\omega_{m}A_{m}(T_{1}, T_{2}) e^{i\omega_{m}T_{0}} \phi_{m}(x) + \\ i\omega_{n}A_{n}(T_{1}, T_{2}) e^{i\omega_{n}T_{0}} \phi_{n}(x) + cc \end{cases}$$
(15)

此处 A_k 表示 k 阶复模态的幅值, cc 表示前面项的 共轭项, 将上式代入二阶近似方程(13)可得

$$D_{0}u_{2} - \nu_{2} = -D_{1}A_{m}e^{i\omega_{m}T_{0}}\phi_{m} - D_{1}A_{n}e^{i\omega_{n}T_{0}}\phi_{n} + cc$$
(16)

$$D_{0}\nu_{2} + Lu_{2} = -i\omega_{m}D_{1}A_{m}e^{i\omega_{m}T_{0}}\phi_{m} - i\omega_{n}D_{1}A_{n}e^{i\omega_{n}T_{0}}\phi_{n} + [A_{m}^{2}e^{2i\omega_{m}T_{0}} + A_{m}\overline{A}_{m}]G_{2}(\phi_{m},\phi_{m}) + [A_{m}A_{n}e^{i(\omega_{m}+\omega_{n})T_{0}} + A_{n}\overline{A}_{m}e^{i(\omega_{n}-\omega_{m})T_{0}}] \times [G_{2}(\phi_{m},\phi_{n}) + G_{2}(\phi_{n},\phi_{m})] + [\overline{A}_{n}e^{2i\omega_{n}T_{0}} + A_{n}\overline{A}_{n}]G_{2}(\phi_{n},\phi_{n}) + cc$$
(17)

外激励通过共振对系统输入能量,假设 $\Omega = \omega_m + \varepsilon^2 \sigma_1$,1:1内共振时 $m \ \pi n$ 阶模态的接近程度 $\omega_n = \omega_m + \varepsilon^2 \sigma_2$,调谐参数 σ_1 既可大于零也可小于零, 而 σ_2 则与内共振形式有关,它在模态转向时间不 改变符号.已有研究^[14]表明1:1内共振解与时间 尺度 T_1 无关,方程(16,17)中 D_1A_m 和 D_1A_n 均为 零,共振项不在二阶近似方程中出现,因此二阶近 似解可表示为

$$u_{2} = A_{m}^{2} e^{2i\omega_{m}T_{0}} \Psi_{mm}(x) + A_{m} \overline{A}_{n} \chi_{mm}(x) + A_{n}^{2} e^{2i\omega_{n}T_{0}} \Psi_{nn}(x) + A_{n} \overline{A}_{n} \chi_{nn}(x) + A_{n} A_{m} e^{i(\omega_{n} + \omega_{m})T_{0}} \Psi_{mn}(x) + A_{n} A_{m} e^{i(\omega_{n} - \omega_{m})T_{0}} \Psi_{mn}(x) + cc$$
(18)

$$\nu_{2} = A_{m}^{2} e^{2i\omega_{m}T_{0}} \eta_{mm}(x) + A_{m} \overline{A}_{m} \zeta_{mm}(x) + A_{n}^{2} e^{2i\omega_{n}T_{0}} \eta_{nn}(x) + A_{n} \overline{A}_{n} \zeta_{nn}(x) + A_{n} A_{m} e^{i(\omega_{n} + \omega_{m})T_{0}} \eta_{mn}(x) + A_{n} A_{m} \xi_{mm}(x) + A_{n} A_{m} e^{i(\omega_{n} - \omega_{m})T_{0}} \eta_{mn}(x) + A_{n} A_{m} e^{i(\omega_{n} - \omega_{m})T_{0}} \zeta_{mn}(x) + Cc$$
(19)

其中 $\Psi_{kk}(\chi_{kk})$ 和 $\eta_{kk}(\zeta_{kk})$ 分别是二阶位移和速度形 函数,将上式代入式(16,17)可得已解耦的位移形 函数的边值微分方程

$$L\Psi_{mm} - 4\omega_m^2 \Psi_{mm} = G_2(\phi_m, \phi_m);$$

$$L\Psi_{nn} - 4\omega_n^2 \Psi_{nn} = G_2(\phi_n, \phi_n)$$
(20)
$$L\Psi_{mn} - (\omega_m + \omega_n)^2 \Psi_{mn} =$$

$$G_2(\phi_m, \phi_n) + G_2(\phi_n, \phi_m)$$
(21)

$$L\chi_{mm} = G_2(\phi_m, \phi_m); \quad L\chi_{nn} = G_2(\phi_n, \phi_n) \quad (22)$$

$$L\chi_{mn} - (\omega_n - \omega_m)^2 / \chi_{mn} =$$

$$G_2(\phi_m, \phi_n) + G_2(\phi_n, \phi_m) \quad (23)$$

边界条件仍为(6)和(7),相应二阶速度场的形函 数为

$$\eta_{mm} = 2i\omega_m \Psi_{mm}; \eta_{nn} = 2i\omega_n \Psi_{nn};$$

$$\eta_{mn} = i(\omega_m + \omega_n) \Psi_{mn};$$
(24)

$$\zeta_{mm} = \zeta_{nn} = 0; \zeta_{mn} = i(\omega_n - \omega_m)\chi_{mn}$$
(23)

将二阶近似解(18),(19)和一阶近似解(15) 代入三阶近似方程,并利用消除久期项条件的可以 得到求解方程如下

$$2i\omega_{m}(D_{2}A_{m} + \mu_{m}A_{m}) = K_{1}A_{n}^{2}A_{n}e^{i\sigma_{2}t} + K_{2}A_{m}^{2}\overline{A}_{n}e^{-i\sigma_{2}t} + 2K_{2}A_{m}\overline{A}_{m}A_{n}e^{i\sigma_{2}t} + K_{3}A_{n}^{2}\overline{A}_{m}e^{2i\sigma_{2}t} + K_{m}A_{m}^{2}\overline{A}_{m}^{2}\overline{A}_{m}^{2} + K_{m}A_{m}A_{n}\overline{A}_{n}^{2}\overline{A}_{m}^{2} - \frac{1}{2}f_{m}e^{i\sigma_{1}t}$$
(26)

$$2i\omega_{n}(D_{2}A_{n} + \mu_{n}A_{n}) = K_{1}A_{n}^{2}A_{m}e^{i\sigma_{2}t} + K_{1}A_{m}A_{n}\overline{A}_{n}e^{-i\sigma_{2}t} + K_{2}A_{m}^{2}\overline{A}_{m}e^{-i\sigma_{2}t} + K_{3}A_{m}^{2}\overline{A}_{n}e^{-2i\sigma_{2}t} + K_{4}A_{m}A_{m}\overline{A}_{m}A_{n} + K_{nn}A_{n}^{2}\overline{A}_{n} - \frac{1}{2}f_{n}e^{i(\sigma_{1}-\sigma_{2})t}$$
(27)

式中 K_{mm} , K_{nn} , K_{mn} 是与共振无关的系数, K_1 , K_2 , K_3 是与共振相关的作用系数,其具体表达式见附录 (B1 - B5)式, $f_k = \langle F, \phi_k \rangle$, $2\mu_k = \langle c, \phi_k^2 \rangle$ (k = m, n),将摄动解 A_k 写为极坐标形式

$$(A_m, A_n) = \frac{1}{2}(a_m, a_n) \exp\{i[\beta_m + \sigma_1 t, \beta_n + (\sigma_1 - \sigma_2)t]\}$$
(28)

其中 *a_j* 和 β_j 均为实函数,将其代入可解性条件 (26)式和(27)式,并将结果分成实部和虚部,整理 之后可得极坐标形式的平均方程

$$\dot{a}_{m} = -\mu_{m}a_{m} - \frac{K_{1}}{8\omega_{m}}a_{n}^{3}\sin\gamma_{1} - \frac{K_{2}}{8\omega_{m}}a_{m}^{2}a_{n}\sin\gamma_{1} - \frac{K_{3}}{8\omega_{m}}a_{m}a_{n}^{2}\sin2\gamma_{1} + \frac{f_{m}}{2\omega_{m}}\sin\beta_{m} \qquad (29a)$$

$$a_{m}\dot{\beta}_{m} = -\sigma_{1}a_{m} - \frac{K_{mm}}{8\omega_{m}}a_{m}^{2} - \frac{K_{mn}}{8\omega_{m}}a_{m}a_{n}^{2} - \frac{K_{1}}{8\omega_{m}}a_{n}^{3}\cos\gamma_{1} - \frac{3K_{2}}{8\omega_{m}}a_{m}^{2}a_{n}\cos\gamma_{1} - \frac{K_{3}}{8\omega_{m}}a_{m}a_{n}^{2}\cos2\gamma_{1} + \frac{f_{m}}{2\omega_{m}}\cos\beta_{m} \qquad (29b)$$

$$\dot{a}_{n} = -\mu_{n}a_{n} + \frac{K_{1}}{8\omega_{n}}a_{n}^{2}a_{m}\sin\gamma_{1} + \frac{K_{2}}{8\omega_{n}}a_{m}^{3}\sin\gamma_{1} + \frac{K_{3}}{8\omega_{n}}a_{m}a_{m}^{2}\sin2\gamma_{1} + \frac{f_{n}}{2\omega_{n}}\sin\beta_{n} \qquad (29c)$$

$$a_{n}\dot{\boldsymbol{\beta}}_{n} = -(\boldsymbol{\sigma}_{1}-\boldsymbol{\sigma}_{2})a_{n} - \frac{K_{nn}}{8\boldsymbol{\omega}_{n}}a_{n}^{3} - \frac{K_{mn}}{8\boldsymbol{\omega}_{n}}a_{n}a_{m}^{2} - \frac{3K_{1}}{8\boldsymbol{\omega}_{n}}a_{n}^{2}a_{m}\cos\boldsymbol{\gamma}_{1} - \frac{K_{2}}{8\boldsymbol{\omega}_{n}}a_{m}^{3}\cos\boldsymbol{\gamma}_{1} - \frac{K_{3}}{8\boldsymbol{\omega}_{n}}a_{n}a_{m}^{2}\cos\boldsymbol{\gamma}_{1} + \frac{f_{n}}{2\boldsymbol{\omega}_{n}}\cos\boldsymbol{\beta}_{n}$$
(29*d*)

式中 $\gamma_1 = \beta_m - \beta_n$.

3 数值分析

在对转动弹簧弹性支撑浅拱的分岔进行数值 研究时,以最低两阶模态(m = 1, n = 2)发生模态转 向1:1内共振附近的非线性响应为研究对象,系统 参数假设为 $k_1 = 10, k_2 = 20, b = 7.0, \mu_m = 0.006, \mu_n$ =0.008, $\omega_m = 51.11, \omega_n = 52.49$,平均方程中的各 个系数如表1所示.

表1 平均方程系数表

Table 1 The coefficients of the averaging equation
--

$\mathbf{K}_{\mathbf{mm}}$	K _{nn}	K _{mn}	K ₁	K ₂	K ₃
1429.58	873.61	653.47	- 55.38	618.69	- 461.37

平均方程的稳态解或固定点可通过在式(29) 中令 $\dot{a}_m = \dot{a}_n = \dot{\beta}_n = 0$ 来获取,在确定稳态解的 响应时一般先选取一个较大或较小的远离共振区 的参数值,再利用 Newton – Raphson 法进行积分得 到一组解,然后利用拟弧长的延拓法^[15]确定整条 曲线.解的稳定性则由对应 4 × 4Jacobian 矩阵的特 征值予以判断,在下面稳态解中实线表示稳定解, 虚线表示不稳定解.

3.1 激励幅值的影响

对于平均方程式(29)中的外激励,设 $f_1 = <$ $F, \phi_1 > , f_2 = 0$ 来分析激励幅值 F 发生变化时系统 的响应,图 5(a),(b)给出了 $\sigma_1 = 0, \sigma_2 = 0$ 和 $\sigma_1 =$ $0, \sigma_2 = 0.10$ 时 a_1 和 a_2 随F增大时的变化规律.在 图 5(a)中可以看到系统的稳态响应值随F的增大 而单调增加,且稳态解一直为稳定解.对于 $\sigma_2 = 0.$ 10的情形,图 5(b)所示激励响应曲线完全呈现另 一番景象,从原点出发的稳定路径在F = 0.77和F = 1.12时分别遇到第1个和第2个 Hopf 分岔点, 两个分岔点之间是不稳定解,随着F的增加稳态解 单调增长至F = 4.59遇到第一个鞍结分岔点SN1, 接下来曲线以不稳定解折回至第3个 Hopf 分岔 点,对应F = 0.89,然后稳态解曲线再经过两个鞍





图 5 激励幅值响应曲线(f₂=0)

Fig. 5 Force - response curves for with $f_2 = 0$

3.2 激励频率的影响

图 6 给出了 $F = 3, f_2 = 0$ 和 $\sigma_2 = 0.05$ 时激励 响应 a_1 和 a_2 随 σ_1 变化的幅频曲线. 由图可知随 着的变化稳态响应的解存在有两条路径,第一条路 径在 σ_1 很小时是稳定解,随 σ_1 的增长稳定性止于 鞍结分岔点 SN1(此时 $\sigma_1 = -0.20$),此过程中 a_2 的幅值非常小,所以结构的非线性响应可以忽略这 一部分的影响,经过 SN1 后稳态解变得不稳定直 至随 σ_1 减小达到另一个鞍结分岔点,然后随着 σ_1 增加稳态解曲线经过两个 Hopf 分岔点 $HB1(\sigma_1 =$ -0.30)和*HB*2($\sigma_1 = -0.21$)达到鞍结分岔点*SN*2 $(\sigma_1 = -0.04)$,两个 Hopf 分岔点之间的稳态解是 不稳定的,由 SN2 出发的不稳定解曲线随 σ_1 的变 化达到第三个 Hopf 分岔点 HB3($\sigma_1 = -0.06$),这 条路径再由 SN3 和另外一个鞍结分岔点最终进入 远离共振区的恒稳状态,此时位移响应以 a_1 为主, a2 的影响可以忽略,这条路径中两阶模态的幅频 曲线均呈双支软弹簧性质,这与平均方程中对幅频 曲线起决定作用的系数 K_{nn}和 K_{nn}的符号相同有 关. 第二条路径是一条闭合路径, 它被两个鞍结分 岔点 SN4(σ₁ = −0.86)和 SN5(σ₁ = −0.04)分为 一条稳定解和一条不稳定解两个部分.





平均方程出现 Hopf 分岔之后,稳态解将变为 周期解,相空间出现极限环,本文利用 Shooting 法^[15]与数值积分求解两点边值问题得到系统的周 期解,并用 Floquet 理论^[15]来判断解的稳定性.图 6 模态响应的幅频曲线中有 3 个 Hopf 分岔点,以此 为起点计算得到整条周期解曲线如图 7(a)所示, 周期解分支中实心圆代表稳定解,空心圆代表不稳 定解.从中可以看到以 HB1 和 HB2 出发的周期解 分支中解是稳定的,随着 σ_2 变化至 – 0.26 时衍生 出另外两条周期解分支,其中一条稳定另外一条不 稳定.从 HB3 出发的周期解分布十分复杂,图 7 (b)所示的周期 T 说明系统除了周期解外还存在 准周期解和混沌解.



3.3 调谐参数的影响

图 8 给出了 $\sigma_1 = -0.1, 0, 0.1$ 三种情况下 a_1 和 a_2 随着调谐参数 σ_2 变化的幅频曲线. 从中可知





σ₁ 取三种不同值时两阶模态的响应曲线在共振区 的主要部分均呈硬弹簧性质;另一方面在远离共振 区系统的2阶模态响应值很小,离共振区越远值越 小,非线性响应中可以忽略这一部分的影响. 三组 稳态解在硬弹簧的下支不稳定解越过一个鞍结分 岔点后均进入一段较短的稳定稳态解,且稳定性都 止于一个 Hopf 分岔点,不稳定解接下来随 σ_2 的增 大增至某一个值时再返回到一个鞍结分岔点,接下 来随着 σ_2 增加则进入远离共振区的恒稳状态,期 间还经过两个 Hopf 分岔点和这两个点之间的一段 不稳定区域. σ_1 取不同值时三组解在整体上分布 规律是一致的,只是表示其特征的具体值大小有差 异.

4 结论

本文研究了浅拱在两端采用转动弹簧支撑的 边界条件下,外激励引起系统出现内共振时的非线 性动力学行为.首先通过研究弹性支撑边界条件下 自振频率及模态的分布规律,发现在一定条件下可 发生模态交叉和转向的内共振.以最低两阶模态之 间的1:1内共振为研究对象,分析了模态转向时的 动力学行为,分析过程中采用多尺度法对动力方程 进行摄动求解,得到极坐标形式的平均方程.通过 外激励幅值、频率以及内部调谐参数对系统响应的 影响研究了系统的稳态解和周期解,结果显示在转 角弹性约束的浅拱中当主要参数处在某一范围之 内时,外激励激发的模态相互作用将导致系统出现 准周期运动和混沌运动.

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RESEARCH ON THE NONLINEAR DYNAMIC BEHAVIORS OF ELASTIC SUPPORT SHALLOW ARCH*

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Abstract The nonlinear dynamic behaviors of an elastic support hinged-hinged uniform shallow arch with two torsional springs located at two ends respectively under external excitation are investigated. Based on the control equation of elastic support shallow arch the multiple scale method was used for the perturbation analysis of internal resonances, and the averaging equation with its polar form was determined. The influence of the stiffness for elastic boundary, which has a one-to-one relationship to the corresponding coefficient in the averaging equation, can be reflected in the natural frequencies and mode shapes of arch structure via the characteristic equation. Then the 1:1 internal resonance between the lowest two modes was selected as studying object for numerical calculation. The analytical results show that there exist internal resonance forms for both cross and veer in the torsional elastic support shallow arch, further the two-mode interaction induced by external excitation may lead to quasi-periodic oscillation and chaotic motion when the systematical parameters are controlled in a certain range.

Key words shallow Arches, torsional elastic support, internal resonance, bifurcation, veer

附录

$$c_{1} + c_{3} = 0$$

$$(A1)$$

$$\omega c_{1} + k_{1} \omega^{1/2} c_{2} - \omega c_{3} + k_{1} \omega^{1/2} c_{4} + k_{2} \pi c_{5} = 0$$

$$(A2)$$

$$\cos \omega^{1/2} c_{1} + \sin \omega^{1/2} c_{2} + \cosh \omega^{1/2} c_{3} + \sinh \omega^{1/2} c_{4} = 0$$

$$(A3)$$

$$(\omega \cos \omega^{1/2} - k_2 \omega^{1/2} \sin \omega^{1/2}) c_1 + (\omega \sin \omega^{1/2} - k_3 \omega^{1/2} \cos \omega^{1/2}) c_2 - (\omega^{1/2} \cosh \omega^{1/2} + k_2 \omega^{1/2} \sinh \omega^{1/2}) c_3 - (\omega^{1/2} \sinh \omega^{1/2} + k_2 \cosh \omega^{1/2}) c_4 + k_2 \pi c_5 = 0$$
(A4)
$$b^2 \pi^3 \omega (\Gamma_1 c_1 + \Gamma_2 c_2 + \Gamma_3 c_3 + \Gamma_4 c_4) + \Gamma_5 c_5 = 0$$
(A5)

其中

$$\begin{split} \Gamma_{1} &= \frac{(1 + \cos\omega^{1/2})}{\pi^{2} - \omega}, \Gamma_{2} = \frac{\sin\omega^{1/2}}{\pi^{2} - \omega}, \Gamma_{3} = -\frac{(1 + \cosh\omega^{1/2})}{\pi^{2} + \omega}, \\ \Gamma_{4} &= -\frac{\sinh\omega^{1/2}}{\pi^{2} + \omega}, \Gamma_{5} = \pi^{4}(1 + \frac{b^{2}}{2}) - \omega^{2} \\ K_{hh} &= \langle \phi_{h}G_{2}(\phi_{h}, \Psi_{hh}) \rangle + \langle \phi_{h}G_{2}(\Psi_{hh}, \phi_{h}) \rangle + \\ 2 \langle \phi_{h}G_{2}(\phi_{h}, \chi_{hh}) \rangle + 2 \langle \phi_{h}G_{2}(\chi_{hh}, \phi_{h}) \rangle + \\ 3 \langle \phi_{h}G_{3}(\phi_{h}, \phi_{h}, \phi_{h}) \rangle \text{ for } h = m, n \end{split}$$
(B1)

$$\begin{split} K_{mn} &= \langle \phi_m G_2(\phi_n, \Psi_{mn}) \rangle + \langle \phi_m G_2(\Psi_{mn}, \phi_n) \rangle + \\ \langle \phi_m G_2(\phi_n, \chi_{nn}) \rangle + \langle \phi_m G_2(\chi_{nn}, \phi_n) \rangle + \\ 2 \langle \phi_m G_2(\phi_m, \chi_{nn}) \rangle + 2 \langle \phi_m G_2(\chi_{nm}, \phi_m) \rangle + \\ 2 \langle \phi_m G_3(\phi_n, \phi_n, \phi_n) \rangle + 2 \langle \phi_m G_3(\phi_n, \phi_m, \phi_n) \rangle + \\ 2 \langle \phi_m G_3(\phi_m, \phi_n, \phi_n) \rangle & (B2) \\ K_1 &= \langle \phi_n G_2(\phi_m, \Psi_{nn}) \rangle + \langle \phi_n G_2(\Psi_{nn}, \phi_m) \rangle + \\ \langle \phi_n G_2(\phi_n, \chi_{mn}) \rangle + \langle \phi_n G_2(\chi_{mn}, \phi_n) \rangle + \\ \langle \phi_n G_3(\phi_n, \phi_n, \phi_n) \rangle + \langle \phi_n G_3(\phi_n, \phi_m, \phi_n) \rangle + \\ \langle \phi_n G_3(\phi_m, \phi_n, \phi_n) \rangle + \langle \phi_n G_2(\Psi_{mm}, \phi_m) \rangle + \\ 2 \langle \phi_n G_2(\phi_m, \chi_{mn}) \rangle + 2 \langle \phi_n G_2(\Psi_{mm}, \phi_m) \rangle + \\ 2 \langle \phi_n G_2(\phi_m, \chi_{mn}) \rangle + 2 \langle \phi_n G_2(\chi_{mm}, \phi_m) \rangle + \\ 3 \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \langle \phi_n G_2(\chi_{mm}, \phi_m) \rangle + \\ \langle \phi_n G_2(\phi_m, \chi_{mn}) \rangle + \langle \phi_n G_2(\chi_{mm}, \phi_m) \rangle + \\ \langle \phi_n G_2(\phi_m, \chi_{mn}) \rangle + \langle \phi_n G_2(\chi_{mm}, \phi_m) \rangle + \\ \langle \phi_n G_2(\phi_m, \chi_{mn}) \rangle + \langle \phi_n G_2(\chi_{mn}, \phi_m) \rangle + \\ \langle \phi_n G_2(\phi_m, \chi_{mn}) \rangle + \langle \phi_n G_2(\chi_{mn}, \phi_m) \rangle + \\ \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \\ \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \\ \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \\ \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \\ \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \\ \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \\ \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \\ \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \\ \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \\ \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \\ \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \\ \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle = \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \\ \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle + \\ \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle = \\ \langle \phi_n G_3(\phi_m, \phi_m, \phi_m) \rangle$$

(B5)

 $\langle \phi_n G_3(\phi_n, \phi_m, \phi_m) \rangle$

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